Development of a Project Cost Estimating Capability

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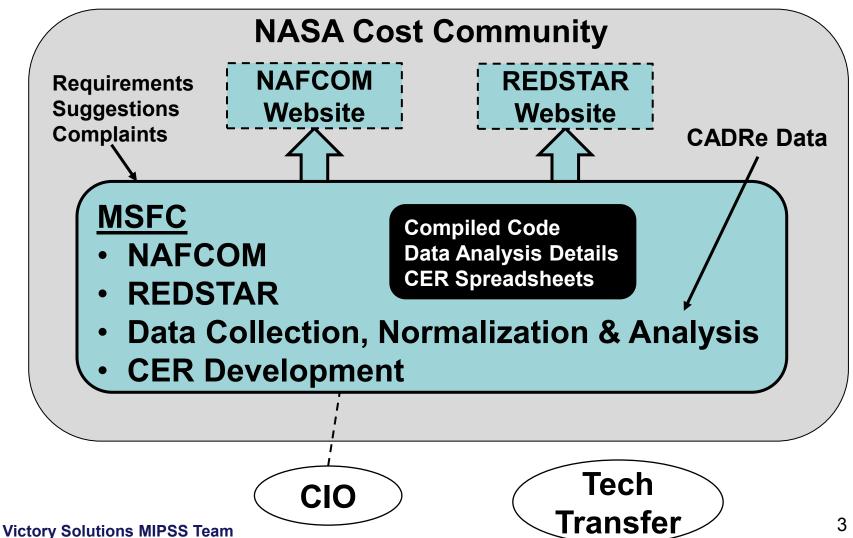
- A New Universe
- Overview of the Requirements and Architecture
- Overview of v1
- The Challenges
- Way Forward







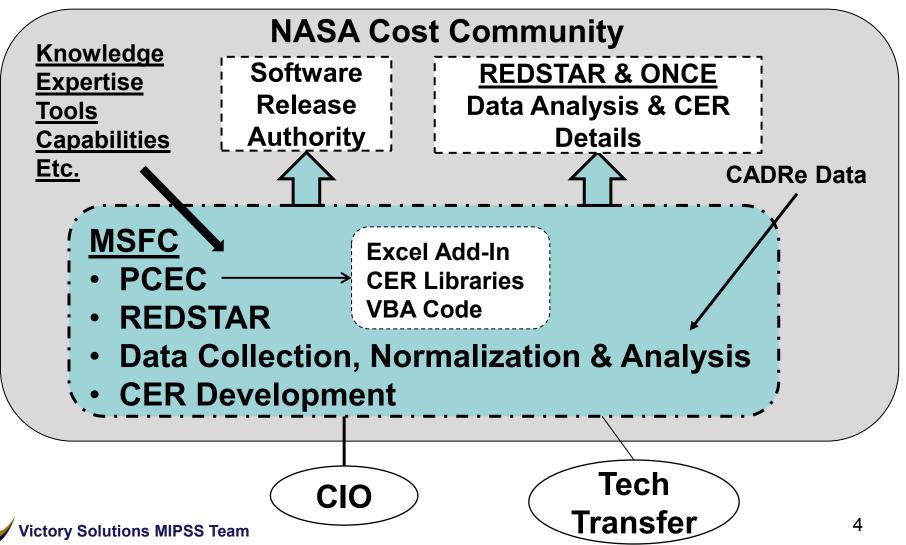
Closed System







Open and Transparent System







- The PCEC team formed a PCEC Steering Committee to guide, define, and develop core requirements
 - Total of 25 Requirements Spread over 3 Levels

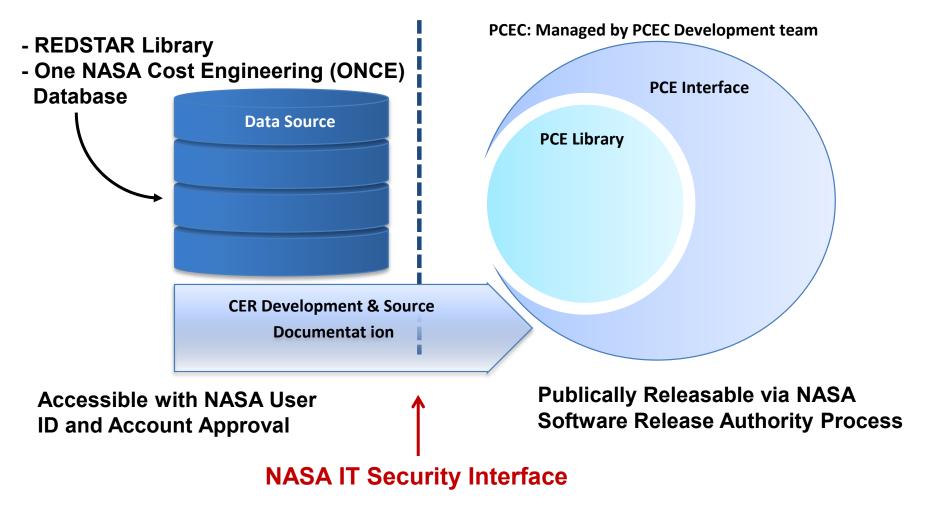
Key Attributes:

- Transparent and Customizable
- Meet all NASA IT Security Requirements
- Include Documentation and Statistics for all CERs
- Traceable to the NASA Standard WBS
- Separate CER Libraries and Software
- Contain No Data or Links to Databases
- Implement an Agreed-Upon List of NAFCOM Capabilities to Carry Forward





Key Elements of Project Cost Estimating Capability (PCEC) – Overview





PCEC Elements



PCEC Library

- Store core cost estimating artifacts (CERs, WBS, Inflation, basic historical mission data)
- Contains NAFCOM12 CERs with associated statistics, as well as NAFCOM WBS templates and other information
- Excel Workbook

				I CLU	Attitud	e com	IOI CLR	Docum	entatio			
Overall PC	EC Attitude Co	ntrol descri	ption:									
The attitud		nent involve								s the process of cor nanding attitude sta		surements taken uipment to carry-out
					De	sign and I	Developm	ent CER				
CER Info	mation											
CER: [DDF	Y12] = 2.608	* [DDMR]^(.734 * [New		12 * [TMI]^-(SpinStab]^-0.				net]^0.559 *	[LV]^-0.437 * [Crew	ed]^1.267 *	
Pre-calcula [AttCo	onDDTech]=[H					+[Computer]	*2	-	-	eter]*1 +[Rendezvo +[PreDevStudy]*0.1		2
Methodok Log-linea UID: 005-01-01	or regression											
	Constant C	DMR	NewDesigr	TMI	WeightPerl	Planet	LV	Crewed	SpinStab	AttConDDTech		
Coefficie	0.959	0.734	1.312	-0.378		0.559						
SE	0.476	0.146	6.66E-02	0.1377293	5.61E-02	0.1045365	0.3478141	0.469568 8.45E-03	0.1473278	0.1001546		

PCEC Interface

- Facilitate the use of the PCEC Library information (e.g., inserting CERs, building WBSs) for creating estimates
- Automate redundant estimating processes
- Excel Add-In

File	Home	Insert Pa	ige Layout	Formu	ılas	Data	R	eview \	∕iew	Dev	eloper
Template	Launch	Estimating Relationships +	CER Details	X Variable Details	Docum Workb		C C	VBS Templat ER Import nflation	• N	2 Ielp	About
Mo	fel		ariable CER	Faireffere		ent		Tools		He	elp
	A1	in Multiva	anable CER	Equation							
4 4 1		FP First Po	ound Cost			F		G	Н		T
2 3	373	SI System	Integratio	n							
4 5		Multiv	ariable CER	Template							
6											
7		Multiv	ariable CFR	Library Sh	eet						
8											
-		201 101			1000						

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CER Library Organization



- Introduction
- General Information
 - Variable List
 - Variable Influence Tables
 - Mission Information
 - WBS Dictionary
 - Inflation Table
 - CER Documentation Help

WBS Templates

- NASA NPR 7120.5E
- NASA CADRe
- NC12 Earth Orbiting Spacecraft
- NC12 Planetary Spacecraft
- NC12 Uncrewed Spacecraft
- NC12 Crewed Spacecraft
- NC12 Launch Vehicle Stage

Recommended CERs

- System Level
- Group Level
- Subsystem Level
- Component Level

Legacy CERs

- Group Level
- Subsystem Level
- Component Level

Contents of Each CER Tab

- CER Documentation
- Cost Calculation
- Risk Calculations





- Microsoft Excel Add-in Workbook (XLAM)
- Provide features to facilitate the integration of project estimating artifacts contained in the PCEC Library into Excel-based cost models
 - Automate Some of the Redundant Processes in Developing a Cost Estimate
 - Custom Tab on Excel's Ribbon
 - VBA based Methods for Formatting and Manipulating Worksheets

The ribbon contains the following button groups:

Models/Estimate

- Load and save model templates
- Launch an Estimate to get stared quickly with a custom model
- Insert individual CERs as a few lines or entire preformatted worksheets

Inform/Document

- Learn more about CERs and variables
- Document and validate CERs used in your workbook

Tools/Help

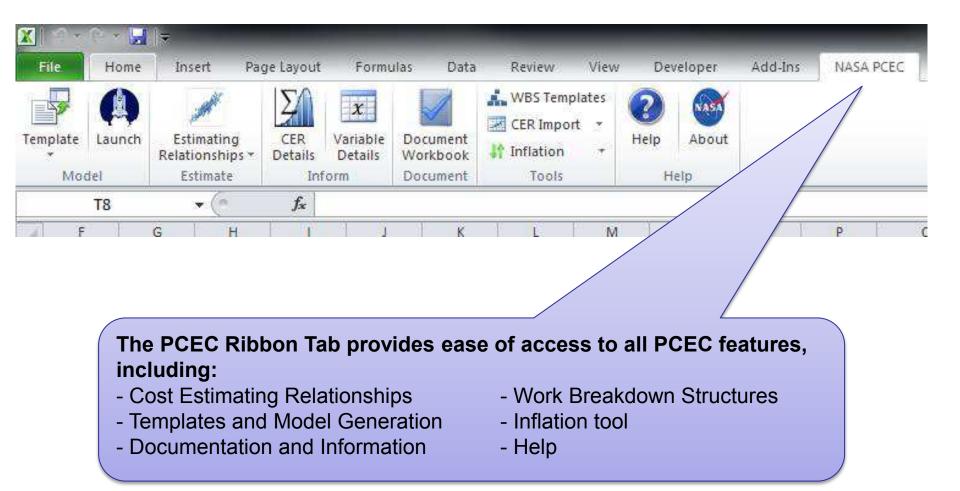
- Insert WBS templates, Library worksheets and Inflation information into your workbook
- Use the in-tool Help file to learn more about what the PCEC can do for you





PCEC v1 Interface Ribbon











- Following an Incremental Development Approach to Incorporate Updates over the Next Several Releases (v1.x)
 - Implementation of Uncertainty for First-Pound and System Integration
 - Integration of Outputs from other Excel-based Models: NICM, SOCM, etc.
 - Phasing
 - User-requested Improvements/Updates
 - Under the Hood Enhancements
- Establishment of the PCEC Review & Release Process
- Upload v1 Data and CER Spreadsheets to REDSTAR
- Development of Training Materials and Example Models with Uncertainty/Risk
- Get State Department Approval for General Release!!



PCEC v2 Challenges



- Integrated Model Construct
 - Integration of Parallel Model Development Efforts
 - Incorporation of Legacy Tools (i.e. NICM)
 - CERs vs. Models
- CADRe vs. Pre-CADRe Data
- Limited Data Sets for Crewed System, Launch Vehicles, and Space Transportation Systems
 - Modeling Development and Production Environments

Approach to Data Analysis

- Full Cost Accounting
- WBS and FBS
- Cost Allocation by Phase
- Objective vs. Subjective Variables
- Modeling System Level Costs





• PCEC Library and Interface v2

- Changes in the Estimating Framework but...
- Same Look, Feel, and Operation as the Existing Interface

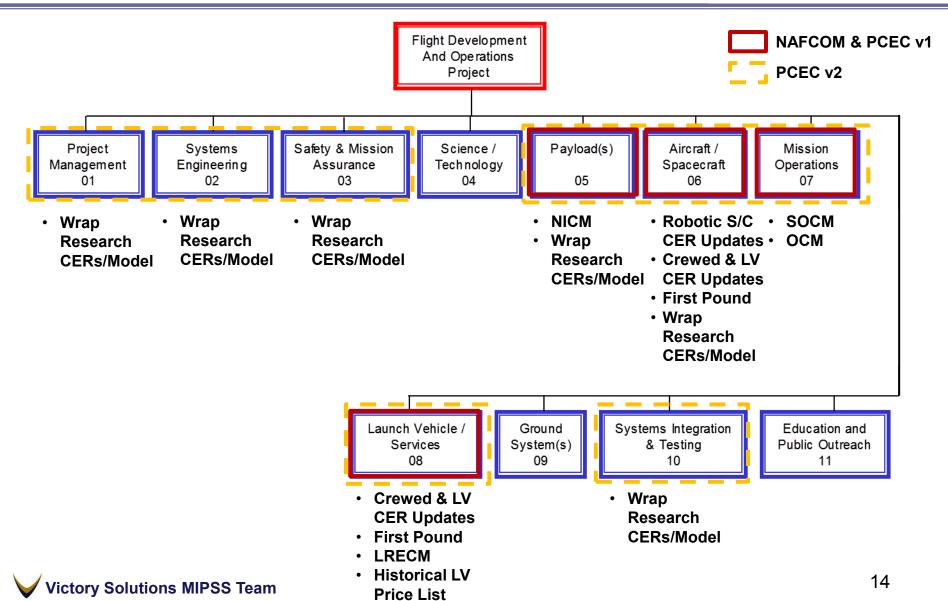
• Key Changes for v2

- NASA Standard WBS: Migration away from NAFCOM WBS
- Space Flight hardware CER updates: New normalizations, new missions added, Tailored CERs/Models by Mission Type (robotic, crewed, launch vehicles)
- New CERs for "wraps": Results of Ongoing PM/SE/MA/I&T Research
- Inclusion of more Models/Capabilities that Enable Total Life-Cycle Cost Estimating
- New Data Normalization/Analysis and CER Workbooks Uploaded to REDSTAR (and ONCE?)
- PCEC v2 is Planned for Release 1Q FY15



PCEC v2 & NASA Std WBS









Questions?















PCEC Library





PCEC CER Library CER Documentation (1 of 3)



		CER Name and description of what is being estimated
Overall PCEC Antenna description: Estimating Level: Component		
The antenna is the part of the uplink/down	nlink that enables a signal to be transmitted and/or received. There are	many different types of antennas, but using one method of
	Design and Development CER	
CER Information		
CER:		Text versions of the 0
	[Deployable]^1.708	and any precalculation
Pre-calculations:		
Methodology:		
Log-linear regression		
UID:		
016-01-01-01		

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PCEC CER Library CER Documentation (2 of 3)



Name Coefficier		WeightPerU 0.490			TMI -1.586	Crewed 1.279			Deployable		Coefficients and
SE	0.922	0.066	0.23336954	0.36861582	0.34973144	0.48633191	0.30590261	0.24181503	0.48992586		associated regressio
P-value	0.000	0.000	9.64E-02		3.24E-05		0.74977497	7.85E-09			information
	ion Statistics gspace	s					Logspace	Unit space			
R^2	0.868		Regression [)F	8	SSR	179.037	5964.2			
Adj R^2	0.849		Residual DF	7	54		27.185				Pagragoion analygia
F-Stat	44.455		Total DF		62		206.222				Regression analysis
F-Stat P-V	-	I				SEE	0.710	_			information and
PING Fact	-				-	Y_Bar	0.129				
SPE	1.06227925				-	CV CV	5.495				statistics
					-	Pearson's Co	0.932	0.81476191			
Max	0.017 91.068 6.190	0.300 155.900 19.741		2.718	4.000 12.000 10.381	1.000 2.718 1.136	0.797	1.000	2.718	4	statistics
Mean Median SD	0.190	5.600 32.004	1.000	1.000	12.000 2.732	1.000 0.468	0.482	0.640	1.000	9.72	7
Median SD	0.999 15.774	5.600 32.004	1.000 0.847 WeightPerU 0.5415	1.000 0.468 CommSC	12.000 2.732	1.000 0.468	0.482 0.160 Crewed 0.5275	0.640 0.249 DDMR 0.2716	1.000		e 8
Median SD	0.999 15.774 ion Matrix DDFY12	5.600 32.004 DDFY12 1.0000	1.000 0.847 WeightPerU 0.5415 1.0000	1.000 0.468 CommSC 0.1109 0.3682	12.000 2.732 ReconSC -0.3090	1.000 0.468 TMI -0.6315	0.482 0.160 Crewed 0.5275 -0.0376	0.640 0.249 DDMR 0.2716	1.000 0.326 NewDesign I 0.6093 0.0339	Deployabl 0.311	e 8 13
Median SD	0.999 15.774 ion Matrix DDFY12 WeightPerU	5.600 32.004 DDFY12 1.0000 0.5415	1.000 0.847 WeightPerU 0.5415 1.0000	1.000 0.468 0.1109 0.3682 1.0000	12.000 2.732 ReconSC -0.3090 -0.2874	1.000 0.468 TMI -0.6315 -0.2452	0.482 0.160 Crewed 0.5275 -0.0376 -0.2381	0.640 0.249 DDMR 0.2716 -0.0633 0.1652	1.000 0.326 NewDesign I 0.6093 0.0339	Deployabl 0.311 0.153 -0.011	e 8 33 5
Median SD	0.999 15.774 ion Matrix DDFY12 WeightPerU CommSC ReconSC TMI	5.600 32.004 DDFY12 1.0000 0.5415 0.1109 -0.3090 -0.6315	1.000 0.847 WeightPerU 0.5415 1.0000 0.3682 -0.2874 -0.2452	1.000 0.468 0.1109 0.3682 1.0000 -0.2381 -0.2488	12.000 2.732 ReconSC -0.3090 -0.2874 -0.2381 1.0000 0.1614	1.000 0.468 TMI -0.6315 -0.2452 -0.2488 0.1614 1.0000	0.482 0.160 Crewed 0.5275 -0.0376 -0.2381 -0.0862 -0.5564	0.640 0.249 DDMR 0.2716 -0.0633 0.1652 -0.2393 -0.2324	1.000 0.326 0.6093 0.0339 -0.0243 0.0652 0.2347	Deployabl 0.311 0.153 -0.011	e 8 33 5
Median SD	0.999 15.774 ion Matrix DDFY12 WeightPerU CommSC ReconSC TMI Crewed	5.600 32.004 DDFY12 1.0000 0.5415 0.1109 -0.3090 -0.6315 0.5275	1.000 0.847 WeightPerU 0.5415 1.0000 0.3682 -0.2874 -0.2452 -0.0376	1.000 0.468 0.1109 0.3682 1.0000 -0.2381 -0.2488 -0.2381	12.000 2.732 ReconSC -0.3090 -0.2874 -0.2381 1.0000 0.1614 -0.0862	1.000 0.468 TMI -0.6315 -0.2452 -0.2488 0.1614 1.0000 -0.5564	0.482 0.160 Crewed 0.5275 -0.0376 -0.2381 -0.0862 -0.5564 1.0000	0.640 0.249 DDMR 0.2716 -0.0633 0.1652 -0.2393 -0.2324 0.3118	1.000 0.326 NewDesign I 0.6093 0.0339 -0.0243 0.0652 -0.2347 0.3682	Deployabl 0.311 0.153 -0.011	e 8 33 5
Median SD	0.999 15.774 ion Matrix DDFY12 WeightPerU CommSC ReconSC TMI Crewed DDMR	5.600 32.004 DDFY12 1.0000 0.5415 0.1109 -0.3090 -0.6315 0.5275 0.2716	1.000 0.847 WeightPerU 0.5415 1.0000 0.3682 -0.2874 -0.2452 -0.0376 -0.0633	1.000 0.468 0.1109 0.3682 1.0000 -0.2381 -0.2488 -0.2381 0.1652	12.000 2.732 ReconSC -0.3090 -0.2874 -0.2381 1.0000 0.1614 -0.0862 -0.2393	1.000 0.468 TMI -0.6315 -0.2452 -0.2488 0.1614 1.0000 -0.5564 -0.2324	0.482 0.160 Crewed 0.5275 -0.0376 -0.2381 -0.0862 -0.5564 1.0000 0.3118	0.640 0.249 DDMR 0.2716 -0.0633 0.1652 -0.2393 -0.2324 0.3118 1.0000	1.000 0.326 0.6093 0.0339 -0.0243 0.0652 -0.2347 0.3682 0.3056	Deployabl 0.311 0.153 -0.011 Da	e B B B B B B B B B B B B B B B B B B B
Median SD	0.999 15.774 ion Matrix DDFY12 WeightPerU CommSC ReconSC TMI Crewed	5.600 32.004 DDFY12 1.0000 0.5415 0.1109 -0.3090 -0.6315 0.5275	1.000 0.847 WeightPerU 0.5415 1.0000 0.3682 -0.2874 -0.2452 -0.0376 -0.0633 0.0339	1.000 0.468 0.1109 0.3682 1.0000 -0.2381 -0.2488 -0.2381 0.1652 -0.0243	12.000 2.732 ReconSC -0.3090 -0.2874 -0.2381 1.0000 0.1614 -0.0862	1.000 0.468 TMI -0.6315 -0.2452 -0.2488 0.1614 1.0000 -0.5564	0.482 0.160 Crewed 0.5275 -0.0376 -0.2381 -0.0862 -0.5564 1.0000 0.3118 0.3682	0.640 0.249 DDMR 0.2716 -0.0633 0.1652 -0.2393 -0.2324 0.3118 1.0000 0.3056	1.000 0.326 0.6093 0.0339 -0.0243 0.0652 	Deployabl 0.311 0.153 -0.011	ata correlation matrix



PCEC CER Library CER Documentation (3 of 3)



AE-3 AMPTE-CCE Apollo LM ATS-6	DMSP-5D DMSP-5D3 DSCS-II DSCS-IIIA	GPSMYP HEAO-1 INTELSAT-IV LRO	NATO III OMV OSO-8 P78		Missions whose da used for regressio			
Chandra CRRES	DSP GPS-IIR	MARISAT Mars Pathfinder	SMS-1 TACSAT					
Variable Descript	ions							
INPUTS:								
<u>Name</u> WeightPerUnit	Brief Desciption Weight Per Unit: V	Veight of each unit (pound	ds)			Possible Values pounds, 0+		
CommSC		pacecraft: Denotes whethe				1 if No, =EXP(1) if Yes.		
ReconSC		Denotes whether the spa		ssance				
TMI		Maturity Index of the elem		Variah	ole description	ons		
Crewed		whether system is Crewee						
NewDesign	~~~~	ng based on the amount of	~~~~~	or a subsystem	-			
Deployable		tes whether the antenna i		duanco manufacturing to shair		0.100% (Decimal in [0.1])		
MfgMethods EngMngt	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	~~~~~	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	dvance manufacturing techniqu management for the system.	es	0-100% (Decimal in [0-1]) 0-100% (Decimal in [0-1])		
FndAvail		ty: Rating that reflects the			0-100% (Decimal in [0-1])			
TestApp		ting based on the level of			0-100% (Decimal in [0-1])			
				······				
IntegComplexity				terfaces involving multiple cont		0-100% (Decimal in [0-1])		
PreDevStudy	Pre-Development	Study: Rating based on th	e amount of the study e	efforts that were/are being cond	ucted prior to the start of	of 0-100% (Decimai in [0-1])		
CALCULATED:								
Name	Brief Desciption					Possible Values		
DDFY12		Development (D&D) Cost	in fiscal vear 2012 millio	ons of dollars (FY12 SM)		FY12 \$M, 0+		
DDMR	······	Rating: Design and Develo				0-100% (Decimal in [0-1])		
	ty Analysis Informa	ation						
Prediction Interval	Distribution Distribution: T-Dist	tribution						
Dogro	es of Freedom: 54							
Degre	c. 51 11 ccu0111. 54							
Squared Design Ma	trix							
Constant	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	.48375 24.9999832 4.999	99664 144.158187 4.99	9999664 -46.202783				
WeightPe	erU 117.48375 375.4	449975 64.4989148 1.61	40661 260.121996 8.31	1594564 -88.30710				
CommSC	24.9999832 64.4	989148 24.9999664	0 54.4618716	Un Un	certainty	<i>information</i>		
ReconSC	4.99999664 1.6	140661 0 4.999	99327 12.4245249	0 -5.0598765				
TMI	144.158187 260.	121996 54.4618716 12.42	45249 337.932302 8.05	5069797 -107.51333				
Crewed	4.99999664 8.31	594564 0	0 8.05069797 4.99	9999327 -1.8516953				
	-46 202783 -88	307181 -16.594409 -5.05	98765 -107.51333 -1.8	3516953 41.2446466 27.8983275	-1.7987906			
DDMR	101202/05 001							
NewDesi	\$		31434 -81.040663	0 27.8983275 30.3028771	L -0.3302715		2	



PCEC CER Library CER Calculation (1 of 2)



	PCEC Antenna Cost Calculation	A calc	ulati	ng versio	on of
Value Name	Input/ Calculation	the CE	R se	et up sim	ilar f
				•	indi (
	Major Inputs	NAFC		2 with	
		dropd	own	lists for	
	Weight	-			
Total Weight		certair	тшр	นเร	
C C					_
	Thruputs				
Design & Development					
System Test Hardware					
Flight Unit					
	Common Multivariable Inputs	DD	FU		
Manufacturing Methods	(3) Mod. Mfg Techniques	5	% 65%	0.5	
Engineering Management	(3) Mod Design Changes	609		0.5	
New Design	(8) New design. Components validated in lab environ or relevant env			1	
Funding Availability	(2) Some Infrequent Delays Possible	5	% 5%	0.5	
Test Approach	(2) Moderate Testing, Qualification at Prototype/Protoflight Level	69	% 15%	0.5	
Integration Complexity	(1) Minimal Major Interfaces Involving Multiple Contractors/Centers	129	% 5%	0.25	
Pre-Development Study	(3) Less Than 9 Months of Pre-Phase C/D Study	129	% 5%	0.75	
	PCEC Antenna Component Specific Inputs				
Deploy	Yes			2.718281828	
Spacecraft Class	(1) Communication			1	
TMI	(12) Technology is mature (> 10 yrs) of flight experience encompassing			12	1



PCEC CER Library CER Calculation (2 of 2)



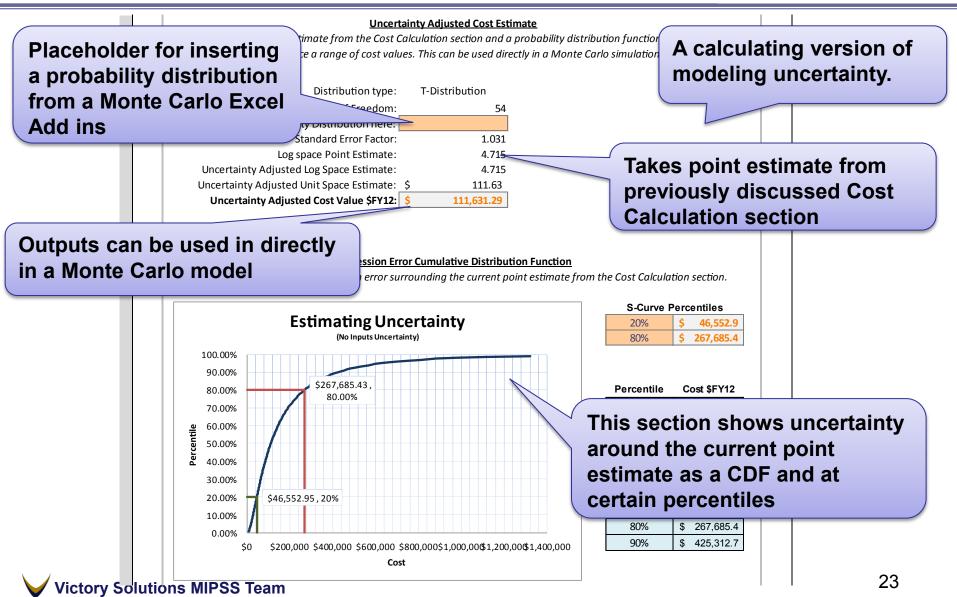
		Other Inputs		
Quantity Next Higher Level			1	1
		CER Methodology		
Multivariable Type	(6) Shuttle Calibrated		Addition	al CER inputs
		Globals		
Flight Year			2015	2015
Units			Thousands	1000
	Syst	tem Test Hardware (STH)		
STH Quantity		· · ·	1	1
FUPercent			130	130
			· · · · · · · · · · · · · · · · · · ·	
DD Point Estimate \$FY12	\$		1	
			CER out	puts
	*			
STH Point Estimate \$FY12	\$			
FU Doint Estimato ÉEV12	Ś		12,915.03	
FU Point Estimate \$FY12	Ş		12,915.03	
Total SEV12	¢		141 225 05	
Total \$FY12	?		141,335.85	

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PCEC CER Library Uncertainty Calculations







PCEC CER Library First Pound Cost (1 of 2)



Overall CER description:		First-Pou	nd Cost CER Documentation (/	Descriptior	n of the First Po	und Cost
	can used to estimate the cost of a gro	oup/subsystem/component using only the v	weight of the element being and	estimating	methodology	
A-value: Calibrated value of the coe	efficient for the individual mission calc	ulated by inputting the mission's actual Cos	st, Weight,	countaing	methodology	
log space.						
b-value : Exponent for the equation (Cost = A * Weight ^ b for the element I	being modeled. It is computed based on reg	rression of historical costs by group/sub	system/component. It represents the	slope of regression line in log space.	
				Calo	culations for eq	uation
Design and Development &		<i>.</i> .		para	ameters (A and	b-values)
CER Information		ons for using				,
	this worl	ksheet				
CER:			1 Innur element weight (and)	uncertainty, if desired) and complexity factor	s in "CER Calcunation at right	
DD FY12 = A * Weight^b) from historical missions in the table	at nght	
$FUFY12 = A * Weight^b$				tion in "Learning, LRIP, and Production Rate (Calculator" cartion at right	
rorriz - A Weight b			4. Link cost results at right to y		D&D Fit Un	it
			- Link cost results at right to y	our was summary wonsheet	Flt Unit A-	
Variable Definitions:					D&D A-Value D&D b-value value Fi	t Unit b-value
A = The geometric mean of the A-values of	f the selected table entries.			Average Values Based on Select	tion N/A N/A N/A	N/A
b = The average of the b-values of the sele	ected table entries				Note: If "N/A" appears in a cell above, please select element with a non-zero value	at least one
Elements Selected	entering an "X" in the "Selected Element" O U Subsystem Element Antenna Antenna Antenna	Column to include that element in the A- and b-v WBS Item Antenna Antenna Antenna	Estimating Level Hardwar Component Uncrewe		ect one or more	nit Hardware ue D D 0 462 0 471 0 465
ACTS	Antenna	Antenna, Multibeam Structure, Multibeam Antenna	Component Uncrewe	analogous d	ata nointe	0 465
ACTS	Automa	Structure, Multibean Antenna	Component Uncrewer	analogous u		438

Component Uncrewed

Amplifier, Traveling Wave Tube



Communication

ACTS

464



PCEC CER Library First Pound Cost (2 of 2)



	First-Pound Cost CER Calculation (Analogy)
Subsystem Name	D&D STH Flight Unit Production Total
CER Calculations	
DD Poir	Iaceholder for input nce For Weight Uncertainty Distribution Distribution Type Parameter 1 Parameter 3 Parameter 4 Parameter 5 Complexity Distribution Distribution Distribution Distribution Distribution Parameter 1 Parameter 2 Parameter 3 Parameter 4 Parameter 5
Fit Unit Point Est \$ Cost Calculations	
Flight Unit Cost LRIP Items Post-LRIP Items	FY12 \$M STH Cost FY12 \$M \$ - \$ - Total \$ - Production Cost FY12 \$M URIP Items \$ - Post-LRIP Item \$ - Post-LRIP Ite
Learning, LRIP, and P	roduction Rate Calculator
Inputs	Q_NHA Q_Prod Unit Slope Theory Type Q_LRIP LRIP Stepdown % Q_STH Flight Unit % (for Production Rate (per year) Production Rate (per year) 1 1 1 100% Unit 0 0% 1 130% 1 100%
Calculations	Learning Multiplier Start Quantity Lot Avg Unit Cost Multiplier (Unit Theory) Lot Avg Unit Cost Multiplier (Cum Avg Theory) Flight Unit Post-RIP Qty 1 0 0.000 1.000 Production Post-RIP Qty 1 1 1.000 1.000 Production Post-RIP Qty 1 0 0.000 1.000 1 0 0.000 1.000
	Production Rate Multiplier 1





PCEC CER Library System Integration (1 of 2)

Scientific

Scientific

Scientific

Earth Orbiting

Earth Orbiting

Earth Orbiting



	System Integr	ation Cost CER Documentation (A	nalogy)	
	System megn			
Overall CER description:	nat can used to estimate the cost of a System Integration	cost element using spacesraft hardware	Description of	f System Integration
	coefficient for the mission calculated by inputting the miss	51 ,	-	
the y-intercept of the regression li		ion's actuarsi cost, weight, and s value	estimating me	ethodology
, , , , ,	on SI Cost = A * HardwareCost ^ b for the element being r	nodeled. Computed based on regression		55
Design and Development	& Flig'			
	Instructions for u	sina		
CER Information				
CER:	this worksheet			
CER: DD FY12 = A * Cost^b			spacecraft costs (D&D, STH, FIt Unit, Prod.) ir ogous mission(s) in the table below	n the "CER Calculations" section to the right (or link to WBS)
$FU FY12 = A * Cost^{A}b$			dditional factors you wish to apply to the co	osts in the section to the right
		4. Link cost re	sults in "System Integration Costs" section at	right to your WBS summary worksheet
Variable Definitions:		Average Valu	es Based on Selection	Spacecraft Cost to Use
A = The geometric m	ulations for aquation	DDT&E A- value	DDT&E b- Flt Unit A- Flt Unit Value value Value	
b = The average of t	ulations for equation	IACO N/A	N/A N/A N/A	
Cost = Spacecraft Ha	meters (A and b-value	STO N/A	N/A	D&D + STH
pulu		GJE IN/A	N/A	D&D + STH
		SE&I N/A PM N/A	N/A N/A N/A N/A N/A N/A	D&D + STH Flt Unit D&D + STH Flt Unit
		LOOS N/A	N/A N/A	
		Note: If "N/A	' appears in a cell above, please select at lea	
Colord and a second				
select one or more entries in the table	by entering an "X" in the "Selected Mission" column to include	that mission in the A- and b-value calculations t	v thai su anuthar	
Elements Selected	0	Tabl	e of hardware t	from Test Operations)
Selected	Hardware			Flt Unit A- Flt U
Mission Mission ACTS	Class Orbit Spacecraft Class Uncrewed Earth Orbiting Communication	Estimating Level which	h to select one	e or more

analogous data points

System

System

System

Uncrewed

Uncrewed

Uncrewed

AE-3

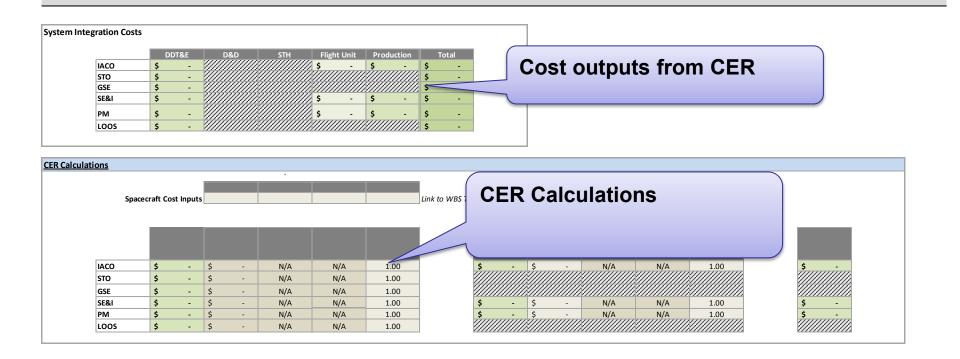
ALEXIS

AEM-HCMM



PCEC CER Library System Integration (2 of 2)











PCEC Interface





PCEC Interface Templates



Structured for CER Inputs CER Output Summary for DDT&E, D&D, STH, Flight Unit, Production and Total Cost

Structured for CER Inputs Multivariable CER FU and DD sections with variable input information and CER formula references in place

DDT&E D&D STH Flight Unit Production	
bolac bab off fight offic fire foundation	Total
\$ 11.021 \$ 11.021 \$ - \$ - \$ - \$	11.021

CER Calculation

 <u>C Power</u> ribution DD	Weight Per Unit	Launch Vehicle	Crewed	New Design	DD Management Rating	Manufacturing Methods	Engineering Management
\$ 11.021	1000	1	1	0.75	0.534	0.8	0.56
	9.8593E-06	1.00602E-05	0.018137938	0.000862701	0.050984275		
	pounds, 0+	1 if No, =EXP(1) if	1 if No, =EXP(1) if	0-100% (Decimal	0-100% (Decimal	0-100% (Decimal	0-100% (Decimal
		Yes.	Yes.	in [0-1])	in [0-1])	in [0-1])	in [0-1])
	Uncertainty Adj.					Degrees of	
	Prediction	SSE Adjusted	Adjustment Factor	SEE	T-Distribution	Freedom	
	11.02053177	0.690063994	1.071588041	0.643963881		31	

Notes:

PCEC Power Distribution FU	Weight Per Unit	Launch Vehicle	Crewed	New Design	Planetary Rating	FU Management Rating	Manufacturing Methods
ş -	1000	1	1	0.75	1	0.738	0.8
	4.66569E-05	0.03926264	0.005373827	0.026690061	0.091531793	0.924792728	
	pounds, 0+	1 if No, =EXP(1) if	1 if No, =EXP(1) if	0-100% (Decimal	1 if No, =EXP(1) if	0-100% (Decimal	0-100% (Decimal
		Yes.	Yes.	in [0-1])	Yes.	in [0-1])	in [0-1])
	Uncertainty Adj.	SSE Adjusted	Adjustment Factor	SEE	T-Distribution	Degrees of	
	#NUM!	#NUM!	#NUM!	0.621478963		30	

Notes:

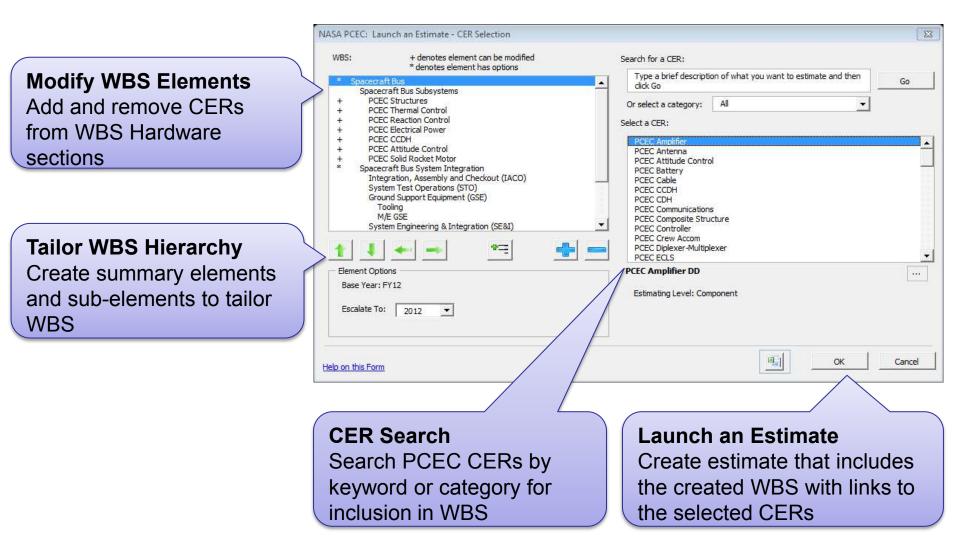
Additional Calculation Sections

Sections for calculating Learning, LRIP, and Production Rate based on CER inputs



PCEC Interface Launch an Estimate







PCEC Interface **Search CER Library**



CER Search Search PCEC CERs by keyword or categories such as: Level, WBS, or Methodology

CER Description See detailed information about the selected CER

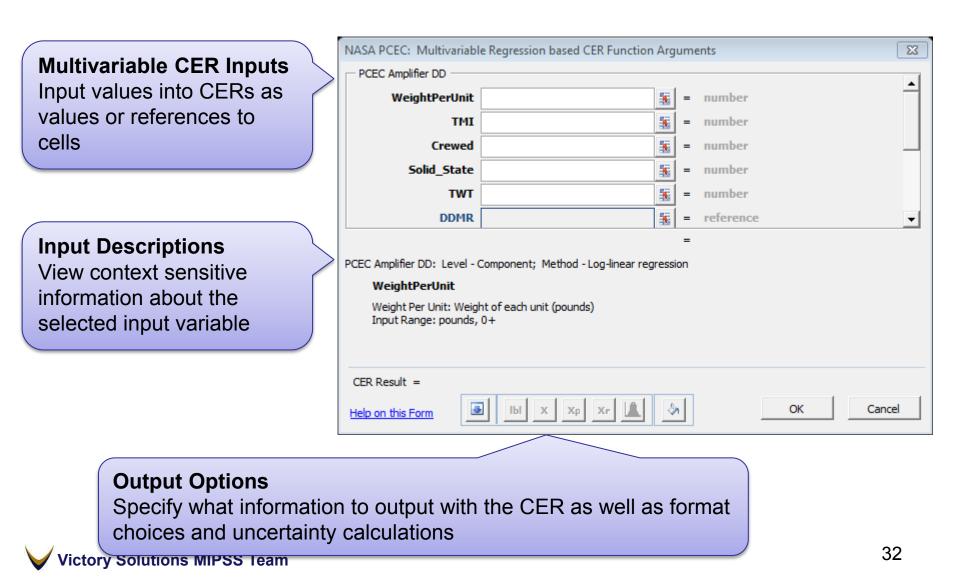
Victory Solutions MIPSS Team

NASA PCEC: Insert CER			23	
Search for a CER:				
Type a brief description then click Go	of what you want to	estimate and	Go	
Or select a category:	Multivariable	•		
Select a CER:				
*Insert a Multivariable Regr	ression equation into the	e selected cell		
PCEC Amplifier DD PCEC Amplifier FU PCEC Antenna DD PCEC Antenna FU PCEC Attitude Control I PCEC Attitude Control I PCEC Battery DD PCEC Battery FU PCEC Cable DD				
PCEC Cable FU PCEC CCDH DD PCEC CCDH FU PCEC CDH DD			-	Multivariable
PCEC Amplifier DD Estimating Level: Comp	onent			CER Info View detailed info about CEF including
Help on this Form		ОК	Cancel	statistics and documentation
	ription the next st your mode	•	ering	31



PCEC Interface Insert Multivariable CER







PCEC Interface CER Details



	NASA PCEC: Multivariable CER Information
CER Search Search PCEC CERs by keyword or category	CER Selection Search for a CER: Type a brief description of what you want to estimate and then click Go Or select a category: All Select a CER: CER Description PCEC Amplifier DD Estimating Level: Component
CER Information View detailed information for Multivariable CERs including: Equation Text, Variable Info, Regression Statistics, Included Missions,	PCEC Amolfier DD PCEC Antenna DD PCEC Antenna DD PCEC Antenna PU PCEC Attitude Control DD PCEC Attitude Control FU PCEC Attitude Control FU PCEC Cable PD PCEC Cable PU CER Information CER Text Variables Regression Missions Correlation SDM CER Text: IDDFY12] = 0.115 * [WeightPerUnit]^1.156 * [TMI]^-0.383 * [Crewed]^1.917 * [Solid_State]^2.823 * [TWT]^0.548 * [DDMR]^0.1 Precalculations: IDDMR] = [MfgMethods]*0.05 + [EngMngt]*0.6 + [FndAvail]*0.05 + [TestApp]*0.06 + [IntegComplexity]*0.12 + [PreDevStudy]*0.12
Variable Correlation, and Uncertainty Info	Help on this Form OK Cancel

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PCEC Interface Variable Information



Variable Information View critical information about CER variables such as Name, Friendly Name, and range of possible input values

Variable Description

View variable definition and a detailed description of how input Values impact a CER output

Variable Name	Friendly Name	Possible Values	
[DDMR]	DD Management Rating	0-100% (Decimal in [0-1])	7
[DDSpecialMat]	DD Special Material Factor	0 if No, 94 if Yes.	
[DDTech]	DD Tech	0+	
[DDTEFY12]	DDTE FY12	FY12 \$M, 0+	ł
[Deployable]	Deployable	1 if No, =EXP(1) if Yes.	
[DesignLife]	Design Life	months, 1+	
[DL120ND79]	DL >= 120 & ND >= 799	1 if No, =EXP(1) if Yes.	
[DLGT72]	Design Life >= 72	1 if No, =EXP(1) if Yes.	
[DRGT1]	Date Rate > 1K	1 if No, =EXP(1) if Yes.	
[EngMngt]	Engineering Management	0-100% (Decimal in [0-1])	
[FndAvail]	Funding Availability	0-100% (Decimal in [0-1])	
[FUFY12]	FU FY12	FY12 \$M, 0+	
[FULouvers_Heaters]	FU Louvers/Heaters	0 for "No Louvers / No Heaters", 3	f
[FUMR]	FU Management Rating	0-100% (Decimal in [0-1])	

This is a dimensionless variable used in the Thermal Control Tech Rating calculation. Special materials or configuration denotes anything out of the ordinary used in the thermal control/protection system, such as the gold plating used on NEAR or the carbon phenolics used on the Galileo Probe. Choices for Special Materials / Special Configurations are the following:

1

(1) Yes = Input of 94

subsystem (D&D)

Help on this Form



Cancel



PCEC Interface Document Workbook



CER Document Table View a table of all CERs included in the workbook. The table contains links to CER Documentation and the CER's cell location in the workbook Estimating Relationship Documentation Report created at: 5/12/2014 3:55:15 PM

Estimating Relationship Name	Worksheet	Address	Valid 🔻
PCEC Amplifier DD	Sheet1	<u>\$B\$2</u>	TRUE
PCEC Structures DD	PCEC Structures CER	<u>\$C\$13</u>	TRUE
PCEC Structures FU	PCEC Structures CER	<u>\$C\$35</u>	TRUE
PCEC Thermal Control DD	PCEC Thermal Control CER	<u>\$C\$13</u>	TRUE
PCEC Thermal Control FU	PCEC Thermal Control CER	<u>\$C\$35</u>	TRUE
PCEC Reaction Control DD	PCEC Reaction Control CER	<u>\$C\$13</u>	TRUE
PCEC Reaction Control FU	PCEC Reaction Control CER	<u>\$C\$35</u>	TRUE
PCEC Electrical Power DD	PCEC Electrical Power CER	<u>\$C\$13</u>	TRUE
PCEC Electrical Power FU	PCEC Electrical Power CER	<u>\$C\$35</u>	TRUE
PCEC CCDH DD	PCEC CCDH CER	<u>\$C\$13</u>	TRUE
PCEC CCDH FU	PCEC CCDH CER	<u>\$C\$35</u>	TRUE
PCEC Attitude Control DD	PCEC Attitude Control CER	<u>\$C\$13</u>	TRUE
PCEC Attitude Control FU	PCEC Attitude Control CER	<u>\$C\$35</u>	TRUE
PCEC Solid Rocket Motor DD	PCEC Solid Rocket Motor CE	R <u>\$C\$13</u>	TRUE
PCEC Solid Rocket Motor FU	PCEC Solid Rocket Motor CE	R <u>\$C\$35</u>	TRUE
PCEC Power Distribution DD	PCEC Power Distribution CER	<u>\$C\$13</u>	TRUE
PCEC Power Distribution FU	PCEC Power Distribution CER	<u>\$C\$35</u>	TRUE

CER Documentation

CER documentation contains the following information about Multivariable CERs: CER Description, Equation Text, Variable Information, Regression Statistics, and Descriptive Statistics,



PCEC Interface Insert WBS



Select WBS		t Work Breakdown Structure
	Select Work Breakdo	
Select WBS from all WBS	NASA NPR 7120.5	E
available in the PCEC	WBS #/Level	WBS Element
	1	Space Flight Project PM
library	1.2	SE
	1.3	Safety & Mission Assurance
	1.4	Science & Technology Payload(s)
	1.6	Spacecraft
WBS View	> 1.7 1.8	Mission operations Launch Vehicle/Services
View the elements of the	1.9	Ground System(s)
selected WBS	1.10	Systems Integration & Testing Education and Public Outreach
Element Description View the elements of the selected WBS	Space Flight Proje	
	Help on this Form	Cancel
		WBS VBS as a new worksheet or as a list es in a specified location
Victory Solutions MIPSS Team		



PCEC Interface Inflation



	•				NASA	NEW	STAR		LATIO)EX(ACTU	ALS 1	HRU	Septe	embe	r 2013	3)		
YEAR.	<u>1959</u>	<u>1960</u>	<u>1961</u>	<u>1962</u>	<u>1963</u>	<u>1964</u>	1	<u>1966</u>	<u>1967</u>		r	<u>1970</u>	1	<u>1972</u>	1 -	<u>1974</u>	1	-	TQ	197
INFL.RATE	4.0%	4.3%	3.2%	4.0%	3.5%	4.5%	3.4%	6.0%	4.9%	5.4%	5.7%	6.9%	6.3%	5.7%	5.7%	7.2%	10.8%	9.0%	2.1%	8.59
FACTORS	1.040	1.043	1.032	1.040	1.035	1.045	1.034	1.060	1.049	1.054	1.057	1.069	1.063	1.057	1.057	1.072	1.108	1.090	1.021	1.08
Inflation Table Insert the NASA NEW START INFLATION INDEX table as a new worksheet in the workbook									ASA PC Destination \$M\$6	n Cell	rt Inflati	on Facto	ors for S	pecific \	/ears	<u></u>				
											Base Y			Start Yea	ar	Fi	inal Year			
			(Infla	ation	Fact	ors				2013	2	•	2014		•	2019	•]	
				Spe	cify a rn a s	rang	je of		rs to		Help on t	his Form				ОК		Cance	9	
Victory	Soluti	ons N	NIPSS	-	tion ta	able													37	



PCEC Interface Insert Library Worksheets



Library Worksheets Generate PCEC Library worksheets as new worksheets in the workbook or as worksheets in a new workbook

NAS	A PCEC: Generate Library Worksheets	8
Selec	ct Library Worksheeets to generate:	
	Available Library Worksheets	
	Variable List	
	Inflation Table	
	First Pound Cost (Analogy)	
	First Pound Cost (Database Average)	
	System Integration (Analogy)	
	System Integration (Database Average)	
	PCEC Amplifier	
	PCEC Antenna	
	PCEC Attitude Control	
	PCEC Battery	
	PCEC Cable	
	PCEC CCDH	
	PCEC CDH	
	PCEC Communications	-
Help	o on this Form OK Cance	el





PCEC Interface In-Tool Help File



