National Transportation Safety Board		NTSB ID	ATL00FA06	9	Aircraft Registration Number: N911AM						
FACTUAL REPORT							Most Critical Injury: Fatal				
ÄYIATION		Occurren	nce Type: Accio	lent	Investigated By: NTSB						
Location/Time					•						
Nearest City/Place	State	ate Zip Code Local Time Time Zone									
SUMNER	GA	3	31789	0230	EDT						
Airport Proximity: Off Airport/Airstrip	Dista	nce From L	Landing Facility:		Direction Fro	m Airpor	t:				
Aircraft Information Summary							•				
Aircraft Manufacturer			Model/Series	8			Type of Aircraft				
Eurocopter			AS350B				Helicopter				
Sightseeing Flight: No		ŀ	Air Medical Tr	ansport Flight: Me	edical Emerger	псу					
Narrative											
Brief narrative statement of facts, conditions and circumstan HISTORY OF FLIGHT	nces perti	inent to the ad	ccident/incident:								
On July 24, 2000, about 0230 eastern daylight time, a Eurocopter AS350B, N911AM, registered to and operated by Critical Care Medflight, was destroyed when it collided with trees and the ground during cruise flight 2 miles east of Sumner, Georgia. The commercial pilot, one flight nurse and company visual flight rules flight plan was filed for the repositioning flight being conducted under Title 14 CFR Part 91. The flight originated from Savannah, Georgia, at 0100, with an intended destination of Sylvester, Georgia. According to the operator, the purpose of the air medical flight operating as LifeFlight 3 was to transport a patient from Coffee County Hospital in Douglas, Georgia, to St. Joseph's Hospital in Savannah, Georgia. After dropping off the patient, the flight was returning to the Worth County Baptist Hospital in Sylvester. At approximately 0227, the pilot contacted the LifeFlight dispatcher and reported a Global Position System (GFS) location 7 miles west/northwest of Tifton, Georgia. No further communication was received from the pilot. Witnesses in the sparsely populated local area stated that at about 0230 they heard the helicopter fly over their homes. They said that it was a dark night, the sky was clear of clouds and that the helicopter fly until out of site then heard a "pop sound" he said it sounded unusual and was not very loud. He woke the next morning and found out about the accident and reported to the local police what he had observed.								ound and nd a cted h an s to l in unty ight ton, pter the the not ocal was			
The flight failed to arrive at its final destination of Sylvester, Georgia. A search was initiated, and the helicopter was located at 0850, 2 miles east of Sumner, Georgia, by a pilot over flying the area. The helicopter collided with trees and came to rest in a swampy, wooded area adjacent to a cotton field.											
PERSONNEL INFORMATION											
The pilot held a commercial pilot certificate for helicopters and in single engine land airplanes, with airplane and helicopter instrument ratings. He also held a flight instructor certificate in helicopters, and an advanced instrument ground instructor certificate. In addition, he possessed a second-class airman medical certificate, dated July 23, 1999, with no waivers or limitations.								e in			
According to company records, Critical Care Medflight hired the pilot on May 11, 2000. The pilot possessed a total of 2,500 hours, 2,280 of which were in helicopters. He had flown 40 hours in the previous 90 days, and 3 hours in the previous 24 hours. His last FAR 61.56 flight review was accomplished on May 5, 2000, in a Eurocopter AS350B.								the			
	I	FACTUA	L REPORT - A	VIATION				Page 1			

National Transportation Safety Board	NTSB ID: ATL00FA069	
FACTUAL REPORT	Occurrence Date: 07/24/2000	
AY LATION ETYBON	Occurrence Type: Accident	
Narrative (Continued)		

AIRCRAFT INFORMATION

The 1980 Eurocopter, AS350B, N911AM, was configured as a medical flight for one pilot and three passengers. The single engine helicopter was owned and operated by C.C. Medflight, Incorporated in Wilmington, Delaware, and operated in Lawrenceville, Georgia. A review of the helicopters logbooks found that on June 7, 2000, the helicopter received an annual and 100 hour inspection with no major discrepancies noted. At the time of the accident the helicopter had accumulated 25 hours of flight time since that inspection.

METEOROLOGICAL INFORMATION

At 0230, weather conditions in Albany, Georgia, located 27 miles to the west, were reported as scattered clouds at 7,500 feet, visibility 7 statute miles, temperature 23 degrees C. (73 degrees F.), dew point 22 degrees C. (72 degrees F.), winds from 150 degrees at 4 knots, and an altimeter setting of 29.93 Hg.

COMMUNICATIONS

At approximately 0227, the pilot contacted the Life-Flight dispatcher and reported a Global Position System (GPS) location 7 miles west/northwest of Tifton, Georgia. No further communication was received from the pilot.

WRECKAGE AND IMPACT INFORMATION

The helicopter impacted the ground on a 330-degree heading. The cabin had separated toward the right side with the right hand forward cross tube mount sliding approximately half the distance from its normal position toward the skid. The forward ends of both skids were buried in the ground with the trailing ends of the skids protruding upward at approximately a 70-degree angle. Impact and the subsequent post-crash fire destroyed the cabin structure. The cockpit and right side of the cabin were thrown to the right of the impact site on a 040-degree heading over a distance of about 150 feet. Flight instruments were found separated from the instrument panel and destroyed. The pilots collective control was found and the stainless steel slide for the friction lock showed that the collective stick was at the maximum pitch setting.

The engine, rotor head and mast, and main gearbox were located in a four-foot deep hole in the approximate correct orientation consistent to the nose of the aircraft as defined by the location of the skids. The tailboom was impaled by the left aft skid spring steel extension. The aft end of the tailboom, vertical fin, tail rotor gearbox, and right horizontal stabilizer were broken away from the tailboom, but located in the general vicinity of the above described wreckage.

Examination of the main rotor shaft, head, and cyclic system remained virtually intact. The swashplate bearing appeared to be functional and could be rotated by hand. The anti-vibrator assembly was ejected from the shaft but remained approximately 10 to 12 inches forward of the rotor head; the attachment bolts were sheared and remained in the threads. The anti-vibrator assembly upper cowling was dented consistent with impact with a solid object. All main rotor blades were intact except that one blade was broken approximately half in length out in flap; there were no impact marks in the leading edge. One main rotor blade appeared to have a leading edge impact at the 1000-mm zone, but with low inertia or with something soft, as the blade remained intact except for trailing edge damage. All three starflex arms were broken in a 45-degree angle from inboard leading edge to outboard trailing edge. The frequency adapters, spherical thrust bearings, and sleeves were all intact except the yellow sleeves were stretched consistent with the impact at the frequency adapter. One pitch change link remained intact, but severely bent; the other two were broken and melted with the rod ends remaining at the attachment points in the blade horns and swashplate. Wood splinters were found in the anti-vibrator assembly and the yellow frequency

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adapter area. The tail rotor gearbox and rotor assembly appeared to be intact. Continuity was assured through the tail gearbox and rotor by rotating the gearbox by hand. The tail rotor pitch links remained intact with no apparent damage to the bearings or links. The tail rotor spider bearing was rotated by hand and appeared to function correctly. The tail rotor driveshaft evidenced no indication of rotation during the crash sequence. No pre-existing impact damage was identified during the examination.

Examination of the engine found it deformed by impact with some parts missing either by the impact or post-crash fire. Missing parts included the bleed valve and start drain valve. The fuel control/governor was only attached by the fuel and airlines. The front case was broken with some of the gearing exposed. The input shaft was broken approximately 2/3 of the distance from the engine toward the main gearbox. After cutting away the case, the aft flex coupling for the input shaft was found to be in good condition. The inlet for the compressor was deformed with the compressor blades bent in two different directions. The tail pipe was severely deformed and was cut away to gain access. The power output coupling had scribe marks installed during assembly of the engine as a reference for over-torque/impact evidence. The marks were found to be misaligned by approximately 1/4 of an inch forcing the gear tighter. The boss around the drive splines was deformed as well. The small coupling area of the power turbine wheel that could be seen through the areas of the tailpipe did not reveal any damage. Very little rotation marks were evident around the turbine wheel.

The engine was crated and sent to Turbomeca Engine Corporation in Grand Prairie, Texas, for further examination. On August 29, 2000, the engine was disassembled and inspected. Present during the examination were representatives from the NTSB, FAA, Critical Care Med-flight, Turbomeca and American Eurocopter. Examination and disassemble of the heavily damaged engine found circular rubs on most of the engines rotating parts, and the twisted power shaft, the fractured/opened muff-coupling and the rotation of the drive pinion into the tightened direction.

MEDICAL AND PATHOLOGICAL INFORMATION

The Georgia Bureau of Investigation Division of Forensic Sciences, in Decatur, Georgia, conducted a post-mortem examination of the pilot. The FAA Toxicology Research Laboratory, in Oklahoma City, Oklahoma, conducted a toxicology examination of the pilot. The examination revealed the following results; Carbon Monoxide Not Performed, Cyanide not performed, Ethanol >>19 Detected in Vitreous, Non-detected in the muscle, >>21 detected in kidney, >>1 N-Propanol detected in Vitreous, >>2 N-Propanol detected in kidney, and >>8 Acetaldehyde detected in kidney. Additionally, the FAA Toxicology Laboratory reported that some of the samples had Putrefaction.

In its investigations, the NTSB routinely requests complete toxicology evaluation on all transportation operators who are fatally injured. In around 10% of such evaluations, ethanol is detected in the blood and/or tissues. In many of these cases, particularly when significant decomposition has taken place, the ethanol present is the result of production of alcohol by microorganisms in the tissues after death, and does not represent the ingestion of alcohol. From the time of the accident until the time of the autopsy approximately 36 hours had passed, allowing sufficient time for microorganisms to enter the tissue.

TEST AND RESEARCH

Advisory Circular 60-4 states in part, "The attitude of an aircraft is generally determined by reference to the natural horizon or other visual references with the surface. If neither horizon or surface reference exist, the attitude of an aircraft must be determined by artificial means from the flight instruments. Sight supported by other senses, allow the pilot to maintain orientation. However during periods of low visibility, the supporting senses sometimes conflict with what is seen. When this happens, a pilot is particularly vulnerable to disorientation. The degree of orientation may vary considerably with individual pilots. Spatial disorientation to a pilot means

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	AVIATION ETYBON	Occurrence Type: Accident	

Narrative (Continued)

simply the inability to tell which way is "up..." Surface references and the natural horizon may at times become obscured, although visibility may be above flight rule minimums. Lack of natural horizon or such reference is common on overwater flights, at night, and specially at night in extremely sparsely populated areas, or in low visibility conditions... The disoriented pilot may place the aircraft in a dangerous attitude... Therefore, the use of flight instruments is essential to maintain proper attitude when encountering any of the elements which may result in spatial disorientation."

ADDITIONAL INFORMATION

In addition to the Federal Aviation Administration, parties to the investigation were American Eurocopter and Turbomeca engines.

The helicopter was released to Phil Powell, Atlanta, Georgia, representing the owners insurance company.

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FACTUAL REPORT		Occu	rrence Dat	e: 07/24/2000							
AVIATION		Осси	rrence Tvr	e: Accident							
Landing Facility/Approach Inform	ation										
Airport Name	Airport ID: Airport Elevation Runway Used Runway Leng						ay Length	n Ru	nway Width		
		Ft	. MSL	0							
Runway Surface Type:											
Runway Surface Condition:											
Type Instrument Approach:											
VFR Approach/Landing:											
Aircraft Information											
Aircraft Manufacturer Eurocopter				el/Series 350B					Serial N 1319	Number	
Airworthiness Certificate(s): Normal											
Landing Gear Type: Skid											
Homebuilt Aircraft? No Nur	Homebuilt Aircraft? No Number of Seats: 4 Certified Max Gross Wt.							LBS	Number	r of Engine	es: 1
Engine Type: Turbo Shaft								Model/Series: ARRIEL 1B			ted Power: I 7 HP
- Aircraft Inspection Information											
Type of Last Inspection			Date of Last Inspection Time Since Last Inspection						Airframe Total Time		
100 Hour			06/2000					25 Ho	ours		4814 Hours
- Emergency Locator Transmitter (ELT)	Information										
ELT Installed? Yes	ELT Operate	ed?			ELT	Aided i	n Locating Ac	cident S	Site?		
Owner/Operator Information											
Registered Aircraft Owner			Stree	t Address 1209 OF	RANGE	E ST.					
C.C. MEDFLIGHT, INC.			City WILMINGTON							State	Zip Code
			Street	Address	GTON					DE	19801
Operator of Aircraft				P.O. BO	X 245						
CRITICAL CARE MEDFLIGHT, IN	City						State GA	Zip Code 30046			
Operator Does Business As: LIFEFLIGHT Operator Designator Code: MFGA											
- Type of U.S. Certificate(s) Held:											
Air Carrier Operating Certificate(s): On-demand Air Taxi											
Operating Certificate:				Operator (Certifica	ate:					
Regulation Flight Conducted Under: Part 91: General Aviation											
Type of Flight Operation Conducted: Po	ositioning										
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_	Z AVIATI													
		A.		Occurrent	ce Type. A	coldeni								
	ot Information					City					Ctoto	Det	o of Dirth	A = 0
Name												e of Birth	Age	
On File						On F	ile				On File	Or	n File	30
Sex: M Seat Occupied: Right Principal Profession: Civilian Pilot Certificate Number: On File														
Certificate(s): Flight Instructor; Commercial														
Airplane Rating(s): Single-engine Land														
Rotorcraft	/Glider/LTA: Helic	-												
Instrument	t Rating(s): Airpl	ane; Helico	pter											
Instructor Rating(s): Helicopter														
Type Ratir	ng/Endorsement fo	or Accident/Ir	ncident Aircra	aft? Yes			С	Current B	iennial Fli	ght R	eview?			
Medical C	ert.: Class 2	Medica	al Cert. Statu	s: Valid Me	dicalno w	/aivers/	/lim.		Date	of La	st Medical	Exan	n: 07/1999	
									I					
- Flight Tir	me Matrix	All A/C	This Make and Model	Airplane Single Engine	Airplane Mult-Engine	Ni	ght	Actual	Instrument Sim	ulated	Rotorcra	ft	Glider	Lighter Than Air
Total Time	e	2500		220							2	280		
Pilot In Co	ommand(PIC)	1800		190							1610			
Instructor						_								
Last 90 Da	-	40	40								_	40		
Last 30 Da	-	12	12								_	12		
Last 24 Ho		3	3				- ·				T	3		
Seatbelt U	Ised? Yes	Shou	Ider Harnes	s Used? Yes			Ioxico	ology Pe	rformed?	Yes		Secor	nd Pilot? No)
	an/Itinerary													
	ight Plan Filed: Co	ompany VF	R											
Departure	Point						State	÷	Airport Ide	entifie	- Dep	parture	e Time	Time Zone
SAVANNAH GA SAV 0100									EDT					
Destinatio	n						State	•	Airport Ide	entifie	r			
SYLVESTER GA SYV														
Type of Clearance: None														
Type of Airspace: Class G														
Weather Information														
Source of Briefing: Flight Service Station; National Weather Service														
Method of Briefing:														
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	Z AVIATION ETYBON			Occurr	e: Accide	ent		1					
Weather	Information				JI								
WOF ID	Observation Time	Time Zone		NOF Ele	vation	WOF	Distance From		dent Site		Direction From	m Accident Site	<u> </u>
					Valion						Diroction rio		- -
ABY	0230	EDT		197	Ft. MSL				27 NM			280 Deg.	Mag.
Sky/Lowes	st Cloud Condition: Sca	attered					7500 Ft. AG	L	Condition of	of Ligł	nt: Night/Dark		
Lowest Ce	iling: None			0	Ft. AGL	Visi	bility:	7	SM	Alti	meter:	29.00	"Hg
Temperatu	ure: 73 °C	Dew Point:		72 °	C Win	d Directio	า: 150			De	nsity Altitude:	1393	Ft.
Wind Spee	ed: 4	Gusts:			Wea	ather Con	tions at Accid	lent S	ite: Visual (Cond	itions		
Visibility (F	RVR): 0 F	t. Visibilit	y (RVV) 0	SM	Intens	ity of Precipita	ation:	Unknown				
Restriction	is to Visibility: None												
Type of Pro	ecipitation: None												
Accident	Information												
Aircraft Da	mage: Destroyed			Aircraft	Fire: Gro	ound			Aircraft Exp	olosio	n None		
Classificati	ion: U.S. Registered/	U.S. Soil											
- Injury Su	mmary Matrix	Fatal	Seriou	ls N	/linor	None	TOTAL						
First Pi	ilot	1					1						
Second	d Pilot												
Studen	t Pilot												
Flight li	nstructor												
Check	Pilot												
Flight E	Engineer												
	Attendants												
Other 0		2					2						
Passer													
	ABOARD -												
Other C		3					3						
	D TOTAL -	0		0 0 0 0 0 3									
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AVIATION											
Administrative Information											
Investigator-In-Charge (IIC)											
BUTCH WILSON											
Additional Persons Participating in This Accident/Incide	ent Investigation:										
RAY BELCHER ATLANTA FSDO ATLANTA, GA 30337											